ustī.	0148N 16 DEC 54	SECRET	3		none.		
25X1 <i>A</i>		and in the second second of the second s Second second	A STATE OF THE STA	13			
re :	DIRECTOR	ROUTING INT	Commission of the responsibility country from the commission of th	[15]	nda is		
from 1			ACCOUNTS OF ANALYSIS REAL PROPERTY OF ANALYSIS AND ANALYSIS ANALYSIS AND ANALYSIS ANALYSIS AND ANALYSIS AND ANALYSIS AND ANALYSIS AND ANALYSIS ANALYSIS ANALYSIS ANALYSIS ANALYSIS ANALYSIS ANALYSIS ANALYSIS ANALYSI	[16]	enta mont		
action:		3 4	PRIORITY				
INFO :		5 6 7	Consumeration to resident and the second	IN 62864			
5X1A	TOR: 0248Z 16 DEC 64	85A 1 - 20		25X1A			
то	PRIORITY	VFO	сіте Г				
	OXCART OPS FLTEST		25X1A	0057ED			
	ARTICLE 129, MAI	DE FLIGHT 79, ON 15 DEC	64.				
	TAKE OFF AT 1321 HOURS FOR I HOUR AND 16 MINUTES. GROSS WEIGHT 121,890						
	LBS, C.G. 20.3 PERCENT, TAKE OFF DISTANCE 7780 FT, TAKE OFF SPEED						
	215 KTS, PRESSURE ALTITUDE 4438 FT, TEMPERATURE 49 DEGREES, WIND CALM,						
	MAXIMUM SPEED 3.14 MACH. MAXIMUM ALTITUDE 78,056 FT, TIME OVER 2.6 MACH						
	THIS FLIGHT 30 MINUTES, TIME OVER 2.6 MACH THIS FLIGHT 20 MINUTES, TIME						
	OVER 3.0 MACH THIS FLIGHT, 15 MINUTES. PURPOSE: CRUISE DATA WITH 450						
	KEAS CLIMB SCHEDULE. Q BAY AND SPECIAL EQUIPMENT: INS.						
	TAKE OFF AND CLIMB ON 458 KEAS LINE WERE NORMAL. REACHED 25X						
	3.1 MACH. 77.000 FT AND 350 KEAS AT PURITY, IDAHO. HE HAD 41.000						
	43,000 LBS OF FUEL BUT COULD NOT REMEMSER TIME FROM TAKE OFF TO THIS						
	POINT. A 180 DEGREE TURN WAS INITIATED AT DILLON, MONTAHA. 25X						
	HAD BEEN INSTRUCTED	vot to adjust power in t	HE TURN AND A	AS RESHLT LOST	,		
	6000 FEET. AS HE LE	VELED OUT, THE LEFT ENGI	NE STALLED FO	CLOVED BY			
	A OROSS INCREASE IN	FUEL FLOW TO 37,000 LBS/	HR. ALSO LEF	LL ENGINE	; ;		
	JSAF review(s) completed.	SECRET	GROUP 1 EXPLUDED FROM ANY NATIC DOWNGRADIN AND DECLASCIFICATE	:	in the second		
	Approved Fo	r Release 2004/05/13 :: CIA-RDP8 ER THAN THE ISSUING OFFICE IS I	9B00980R0003000 PROHIBITED. com	060055-6 Z			

	SECPEI				
25X′	1A IN 62864	PAGE TWO			
25X´	EGT WENT TO 940 DEGREES, AND REMAINED THE		4		
	THE EGT DID NOT COME DOWN UNTIL POWER WAS	REDUCED TO ONE HALF MIL POWER.			
	THEN ABORTED TEST AND	LEFT ENGINE FUEL FLOW 2	25X1		
	WAS 10,000 LBS AT IDLE AND 4000 LBS WITH THE FNGINE SHUT DOWN. GEAR				
	EXTENSION ON EMERGENCY SYSTEM WAS NORMAL AS WAS CHUTE DEPLOYMENT. THE				
	INS OPERATION APPEARED TO BE GOOD.	HAD SAS PITCH A, B AND M 25	X1A		
	LIGHTS ON INTERMITTENTLY THROUGHOUT THE F	LT.	,		
	END OF MEGGAGE				